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## Generation-X: An X-ray observatory designed to observe first light objects

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### 12 Abstract

The new cosmological frontier will be the study of the very first stars, galaxies and black holes in the early Universe. These objects are invisible to the current generation of X-ray telescopes, such as Chandra. In response, the Generation-X (“Gen-X”) Vision Mission has been proposed as a future X-ray observatory which will be capable of detecting the earliest objects. X-ray imaging and spectroscopy of such faint objects demands a large collecting area and high angular resolution. The Gen-X mission plans 100 m<sup>2</sup> collecting area at 1 keV (1000× that of Chandra), and with an angular resolution of 0.1”. The Gen-X mission will operate at Sun–Earth L2, and might involve four 8 m diameter telescopes or even a single 20 m diameter telescope. To achieve the required effective area with reasonable mass, very lightweight grazing incidence X-ray optics must be developed, having an areal density 100× lower than in Chandra, with mirrors as thin as 0.1 mm requiring active on-orbit figure control. The suite of available detectors for Gen-X should include a large-area high resolution imager, a cryogenic imaging spectrometer, and a grating spectrometer. We discuss use of Gen-X to observe the birth of the first black holes, stars and galaxies, and trace their cosmic evolution.

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**Keywords:** Gen-X; Vision Mission; X-ray astronomy; Active optics; Lightweight optics

### 26 Contents

1. Introduction . . . . .	00
2. Summary of science goals of Generation-X . . . . .	00
2.1. Observe the birth of the first black holes, stars and galaxies . . . . .	00
2.2. Trace the evolution of black holes, galaxies and the elements they produce from the earliest times to the present epoch . . . . .	00
3. Generation-X mission concept . . . . .	00
4. Generation-X mission architecture . . . . .	00
4.1. Architecture trades . . . . .	00
4.1.1. Constellation definition . . . . .	00
4.1.2. Launch vehicle . . . . .	00
4.1.3. Orbit . . . . .	00
4.2. Spacecraft configuration . . . . .	00

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38	5.	Generation-X mission technology . . . . .	00
39	5.1.	Telescopes and optics . . . . .	00
40	5.2.	Configuration . . . . .	00
41	5.3.	Materials and fabrication . . . . .	00
42	5.4.	Alignment and figure control . . . . .	00
43	6.	Instruments . . . . .	00
44	6.1.	A cryogenic imaging spectrometer . . . . .	00
45	6.2.	A grating spectrometer . . . . .	00
46	6.3.	Imager . . . . .	00
47		References . . . . .	00
48			
49			

## 50 1. Introduction

51 Astronomers are always in search of more photons, to  
 52 study ever fainter objects in ever finer detail. In this way, they  
 53 have pushed the frontiers of astrophysics back to the time  
 54 when the Universe was only a small fraction of its present  
 55 age. The new frontiers are to find the very first stars, black  
 56 holes and galaxies to form in the universe. This requires ever  
 57 larger telescopes, with ever finer angular resolution,  
 58 throughout the electromagnetic spectrum. In the optical  
 59 and infrared efforts are underway to build telescope mirrors  
 60 up to 20, 30 or even 100 m in diameter, with active and adap-  
 61 tive optics control to achieve resolution near or at the diffrac-  
 62 tion limit. In the radio, the Square Kilometer Array  
 63 interferometer is being planned. Similarly, a large X-ray tele-  
 64 scope in space could observe the formation of the first galax-  
 65 ies and black holes in the Universe. To this end, the X-ray  
 66 astronomical community conceived the large-area high-res-  
 67 olution Gen-X mission, which would succeed the Constella-  
 68 tion-X (Con-X) mission, and outlined a baseline concept for  
 69 Gen-X (Cameron et al., 2004) which was successfully pro-  
 70 posed for a one year study as a NASA Vision Mission in  
 71 2004. We report the status of that study, which is developing  
 72 and refining the science objectives for Gen-X, concepts for  
 73 mission architecture, and the key technologies needed for  
 74 the X-ray telescope and instruments. In the following sec-  
 75 tions, we discuss some of the far-reaching science goals,  
 76 and then present our areas of study and outline some of  
 77 the technical issues being addressed in each area.

## 78 2. Summary of science goals of Generation-X

79 Gen-X will be able to address a wide variety of key  
 80 goals. The two goals most relevant for this Workshop are  
 81 discussed briefly here.

### 82 2.1. Observe the birth of the first black holes, stars and 83 galaxies

84 The exact sequence of structure formation after the  
 85 CBR was released at  $z \approx 1100$  and before the highest  
 86 known redshifts quasars and galaxies at  $z \approx 6$  is currently  
 87 mostly a theoretical frontier. Since we know massive black

88 holes (BHs) exist at  $z \approx 6$  (age  $\approx 1$  Gyr), the first stars, gal-  
 89 axies, and BHs must have formed much earlier, and the  
 90 BH-accretion must be rapid, releasing X-rays in the pro-  
 91 cess. Recent cosmic background polarization results from  
 92 WMAP (Spergel et al., 2003) put this first epoch of energy  
 93 injection at  $z \approx 10\text{--}20$  (age  $\approx 0.5\text{--}0.2$  Gyr). The first gener-  
 94 ation of stars are likely to be massive ( $\gtrsim 100 M_{\odot}$ ), and  
 95 these stars will quickly burn their nuclear fuel collapsing  
 96 to form the first BHs, perhaps in ‘pair instability’ super-  
 97 nova (SN) explosions 100 $\times$  more powerful than SN today  
 98 (Heger and Woosley, 2002). These SN may be visible as  
 99 Gamma-ray bursts (GRBs; Loeb, 2005) detectable by other  
 100 satellites simultaneously. Gen-X can obtain spectra of their  
 101 X-ray afterglows, determining GRB distances and total  
 102 energy output, and probing the high-redshift intergalactic  
 103 medium (IGM).

104 Early BHs (Wyithe and Loeb, 2003) of  $\sim 500 M_{\odot}$  at  
 105  $z \approx 15$  can be detected at their Eddington luminosity by  
 106 Gen-X at a flux of  $3 \times 10^{-20}$  cgs, or 1000 $\times$  fainter than  
 107 Chandra can reach. The 0.1–10 keV Gen-X band is  
 108 uniquely tailored to the detection of these first BHs. Chan-  
 109 dra deep surveys (Alexander et al., 2003) find 10 $\times$  more  
 110 active galactic nuclei (AGN) per unit solid angle than other  
 111 wave-bands. Most high energy radiation from quasars is  
 112 emitted below 100 keV, which at  $z \approx 15$  will be seen below  
 113 7 keV. The 1.5 keV radiation can escape through the  
 114 enshrouding dust and gas expected in these primordial  
 115 objects, and will be redshifted to  $\sim 0.1$  keV. Gen-X will thus  
 116 detect such BHs, and measure the accretion-powered lumi-  
 117 nosity of the high-redshift Universe.

118 Galaxy evolution has been most dramatic at  $z \approx 1\text{--}3$   
 119 (ages  $\approx 6\text{--}2$  Gyr), causing the broad peak in the cosmic  
 120 SFR during that epoch. These galaxies should have several  
 121 young SN remnants (SNR) and “ultra-luminous X-ray  
 122 sources” (ULXs; York et al., 2003), which might be inter-  
 123 mediate mass BHs (several  $1000 M_{\odot}$ ). Gen-X can detect  
 124 both at their expected flux of  $\gtrsim 10^{-19}$  cgs. HST shows that  
 125 at such redshifts galaxies are  $\sim 0''.5$  across, requiring the  
 126 low background rate specified in Table 1 for detection.  
 127 Detecting SNRs, ULXs, and low luminosity AGN  
 128 (super-massive black holes, SMBHs) against the unre-  
 129 solved X-ray binary (XRB) sources and the hot interstellar  
 130 medium (ISM) in these objects requires an angular resolu-

Table 1  
Key mission parameters for Gen-X

Parameter	Baseline	Study range
Effective area (m <sup>2</sup> )	100	50–150
Resolution (HPD)	0".1	0".1–1".0
Energy resolution	$E/dE = 1000$	$10^3$ – $10^4$ @1 keV
Background (0.5–2.0 keV) (cts/ks/arcsec <sup>2</sup> )	0.004	0.04–0.004
Energy range (keV)	0.1–10	0.05–10
Field of view	5'	5–15'
Time resolution (μs)	50	10–100
Count rate limit (cts/read)	0.05	0.05–0.5
Sky availability (%)	90	70–90
Calibration	3% absolute	3–10% absolute

131 tion  $\sim 0".1$ . Efficient mapping requires a FOV of at least 5',  
132 which matches that of JWST.

133 Gen-X will explore galaxies from high redshift to the  
134 present, studying the X-ray evolution of their components  
135 with cosmic epoch. The cosmic star formation rate (SFR)  
136 was 10–100× higher at  $z \approx 1$ –3 than at present. Since X-  
137 ray luminosities in 'normal' galaxies are well correlated with  
138 the SFR (Grimm et al., 2003), X-ray evolution with redshift  
139 is expected (White and Ghosh, 1998). This is confirmed cru-  
140 dely by the Chandra 'stacking' detection of the integrated  
141 emission of large samples of 'Lyman break' galaxies at  
142  $z \approx 2$ –3 (Brandt et al., 2001a,b; Nandra et al., 2002), which  
143 reveals higher X-ray luminosities at higher redshifts. Gen-X  
144 will study hundreds of these galaxies, obtaining  
145  $\sim 400$  counts in  $10^6$  s for a  $z \approx 3$  galaxy (Fig. 1), spatially  
146 separate the integrated XRB emission from a nuclear BH,  
147 and spectrally separate the hard XRB emission from hot  
148 gas (e.g., Fabbiano, 1989; David et al., 1992), revealing  
149 the true SFR, even in dust enshrouded protogalaxies.

150 2.2. Trace the evolution of black holes, galaxies and the  
151 elements they produce from the earliest times to the present  
152 epoch

153 A recent surprise is the discovery that the mass of the  
154 SMBHs in the centers of nearby normal galaxies and the

stellar velocity dispersion of their stars are tightly linked. 155  
This  $M_{\text{BH}}-\sigma_v$  relation (e.g., Gebhardt et al., 2000) suggests 156  
that SMBHs and galaxies evolve in a tightly coupled fash- 157  
ion. However, there is little or no information on the nat- 158  
ure of this relation at higher redshifts. In protogalaxies 159  
the SMBH may well be proportionately larger (Wyithe 160  
and Loeb, 2003). At  $z \approx 10$ , a young galaxy of mass 161  
 $\sim 10^8 M_\odot$  could harbor a  $10^6 M_\odot$  SMBH. Such an SMBH 162  
accreting at the Eddington limit will have an X-ray flux of 163  
 $10^{-16}$  cgs, providing  $\sim 10^4$  counts for spectral analysis in 164  
 $10^6$  s with Gen-X, plausibly yielding the galaxy redshift 165  
via the iron line (redshifted to  $\sim 0.5$  keV). To maintain 166  
the  $M_{\text{BH}}-\sigma_v$  relation, galaxy mergers must merge their 167  
BHs, triggering AGN activity (Hernquist and Springel, 168  
2003) at almost all wavelengths. Gen-X is designed to 169  
observe all these objects in detail. 170

### 3. Generation-X mission concept 171

Table 1 lists the key parameters for Gen-X. The effective 172  
area of 100 m<sup>2</sup> at 1 keV allows detection of a distant black 173  
hole in  $10^6$  s. The angular resolution of 0".1 HPD results in 174  
negligible background in that time. Achieving both the 175  
large area and fine angular resolution will require break- 176  
throughs in technology for X-ray telescopes. The baseline 177  
concept being studied consists of four identical telescopes, 178  
each having an 8 m diameter mirror assembly, attached to 179  
an instrument platform at a 50 m focal length. An alterna- 180  
tive configuration consists of a single telescope, with a mir- 181  
ror assembly of about 20 m diameter having 100 m<sup>2</sup> 182  
effective area at 1 keV, and with focal lengths in the range 183  
80–125 m. A detached, formation flying spacecraft would 184  
carry focal plane instruments. Table 2 compares the base- 185  
line Gen-X telescope parameters with those of previous 186  
or planned missions. In the four-telescope concept, each 187  
telescope would be separately launched on an expendable 188  
launch vehicle. Each telescope will be carried as six seg- 189  
ments to fit in an existing large payload fairing, and with 190  
a deployable attachment to the focal plane assembly. Each 191  
telescope will have an instrument suite consisting of a grat- 192

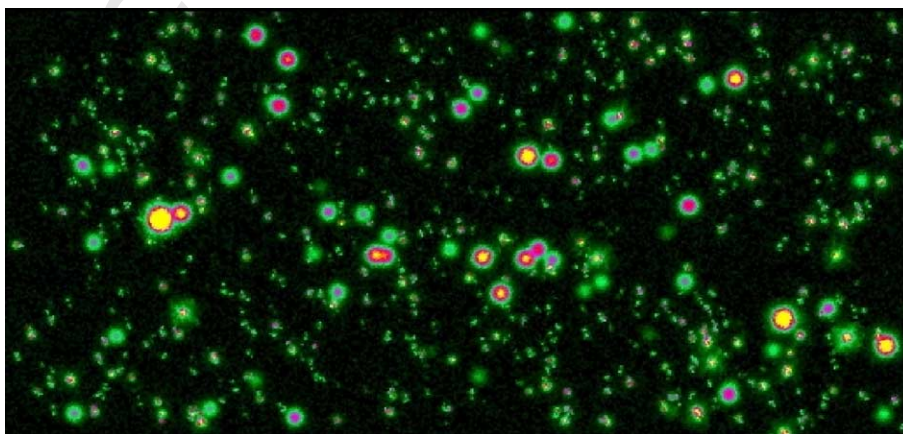


Fig. 1. Slice of a simulated  $10^6$  s Gen-X image of the Hubble deep field (HDF). Chandra saw 17 sources in the same exposure time (Brandt et al., 2001a,b). With realistic fluxes and sizes, most of the 3000 HDF galaxies will be detected with Gen-X, and over 800 will have  $\geq 400$  counts.

Table 2  
Baseline Gen-X telescope parameters

Mission	Telesc. modules	Angular HPD (")	Module effective area (m <sup>2</sup> )	Module mass (kg)	Mirror technology
Chandra	1	0.5	0.1	1000	Zerodur shells
XMM-Newton	3	15	0.15	420	Replicated Ni shells
Astro-E 2	5	90	0.04	12	Replicated Al segments
Constellation-X	4	15 (Goal = 5)	0.75	700	Thermally formed or replicated segments
Generation-X	4	0.1	25	3000	Thin nested segments with active control

ing spectrometer, micro-calorimeter imaging spectrometer, and an active pixel large format imager. A single-telescope implementation of Gen-X would require multiple launches followed by in-flight assembly. A focal plane with the same three instruments would be launched as a separate vehicle.

## 4. Generation-X mission architecture

### 4.1. Architecture trades

We are carrying out mission architecture trades, associated with the different telescope configurations described above, to refine our concept for Gen-X. These trades include the amount of ground vs. on-orbit assembly of the optics, and separate or combined vehicles for the optics and focal plane detectors. The focal plane architecture must accommodate a variety of instruments. Furthermore, since the telescope will certainly remain state of the art for much longer than an instrument lifetime, Gen-X must provide for the replacement or refurbishment of instruments detectors at the L2 location. If multiple telescopes and spacecraft are utilized to achieve the full mirror area, the instruments might be arranged in a series of identical focal planes, or with unique instruments dedicated to individual telescopes. Several issues influence such decisions, including the payload mass and size capability of the launch vehicle, possibilities for astronaut-assisted or robotic on-orbit assembly which might be developed in the future, vehicle propulsion, and formation-flying capabilities.

#### 4.1.1. Constellation definition

While the four-telescope configuration was the starting point of the study, we have identified promising single-telescope configurations. Several factors will influence the trade-off between different configurations. The key driver is the expected performance of the assembled optics, including manufacturing tolerances and diffraction effects (Reid et al., 2004).

#### 4.1.2. Launch vehicle

Currently, Delta 4H and Atlas V vehicles could be used to launch Gen-X, but we anticipate advances in launch weight and propulsion technologies, as well as possible increases in payload size envelopes that future very large space telescopes may require. Gen-X must achieve about an order of magnitude improvement in effective area to weight ratio relative to Con-X. This is less than the improvement from Chandra to Con-X,

and is a reasonable goal. For the four telescope option, we budget a baseline optic-module weight of 3000 kg per telescope.

#### 4.1.3. Orbit

A pre-requisite to mission definition is orbit selection. For the large telescopes required by Gen-X, a low-disturbance orbit is needed, to minimize structural mass and/or station-keeping fuel or power, and thermal stresses. This eliminates low earth orbit (LEO) or even the highly elliptical Chandra or XMM type-orbits, and favors Sun–Earth L2. We might use astronauts and/or robotics to assist in telescope assembly and alignment at LEO or at the Earth–Moon L1, and then travel to the Sun–Earth L2. For Gen-X, we will draw on the demonstrated or planned capabilities of other upcoming missions such as Con-X, JWST, LISA, and TPF, to make such architecture and orbit decisions.

### 4.2. Spacecraft configuration

Gen-X presents significant challenges for the spacecraft platform for telescope(s) with such large mirror area, long focal length and high count rates. For telescopes with 50–200 m focal length, the optics and detector assemblies require either attachment via a deployed boom, or separate formation flying vehicles. The resulting position control tolerances between optics and detector are of order 0.1–0.3 mm in separation, and several mm in the plane perpendicular to the optical axis, which are modest compared to the expected formation flying capabilities. Our trade study is considering the thermal stability required to maintain telescope and instrument performance, including the possibility that low temperature operation might be needed to limit heater power requirements for the optics. Angular momentum management for pointing and slewing is also an issue.

## 5. Generation-X mission technology

### 5.1. Telescopes and optics

Efficient X-ray reflection in the broad energy range of Gen-X requires grazing incidence optics. The baseline optics for Gen-X uses the Wolter I geometry, consisting of concentric nested mirror shell paraboloid and hyperboloid pairs. Table 3 shows a representative Wolter I configuration for a four telescope and for a single-telescope

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Table 3  
Representative Wolter optics configurations

Mission configuration	Outer mirror diameter (m)	Inner mirror diameter (m)	Focal length (m)	Mirror shell length (m)	Number of mirror pairs	Field of view (arcmin)	Effective area @ 1 keV (m <sup>2</sup> )
Four telescope	8.0	1.6	50	1.0	260	4.0	25
Four telescope	17.0	4.0	80	1.0	400	4.0	100

278 implementation. Fig. 2 shows the effective area of those  
 279 optics configurations, assuming gold reflecting surfaces.  
 280 In addition to Wolter optics, which provides optimal on-  
 281 axis resolution, we have considered polynomial shaped  
 282 optics, which provide better off-axis resolution at the cost  
 283 of degraded on-axis resolution. The field of view for Gen-  
 284 X must compromise between the need to fully cover nearby  
 285 extended astronomical targets and match other observato-  
 286 ries such as the planned JWST deep surveys for First Light  
 287 objects (see Windhorst et al., this volume), and the practi-  
 288 cal limitations for X-ray detectors having focal plate-scales  
 289 of about 2"/mm.

290 The theoretical angular resolution of our configurations  
 291 easily meets the 0".1 requirement for Gen-X, so that dif-  
 292 fraction effects and manufacturing tolerances dominate  
 293 the achievable angular resolution (Reid et al., 2004). In  
 294 general, larger diameter optics have larger grazing angles,  
 295 and tighter mirror figure control is needed to achieve max-  
 296 imum encircled energy within the desired angular resolu-  
 297 tion. The two reflections in the Wolter I optical geometry  
 298 means that figure requirements for each optic scale as the  
 299 square root of the system performance. Gen-X thus  
 300 requires an improvement factor of 7× over the Con-X  
 301 goals, and 3× better than achieved for Chandra.

302 Although the effective area is 30× larger than of Con-X,  
 303 the weight cannot be more than a few times larger. The  
 304 hundreds of mirror pairs shown in Table 3 imply that each  
 305 must be extremely thin. We therefore consider an entirely  
 306 new approach for grazing incidence X-ray telescopes:  
 307 adjusting the mirror alignment and figure on-orbit. To  
 308 derive the technology to be applied, we have considered  
 309 the following issues: What materials should be used, what  
 310 level of precision can be attained by manufacturing toler-

ances and material stability, what amount of adjustability  
 must be provided, how are on-orbit adjustments made,  
 and what metrology, calibration techniques, and algo-  
 rithms are used in flight to evaluate and correct the mir-  
 rors? We have been performing finite element mechanical  
 analyses to evaluate requirements for initial on-orbit align-  
 ment and for on-orbit adjustment methods.

## 5.2. Configuration

Achieving high angular resolution with a very large area  
 is the key feature of the Gen-X mission. One could list  
 many approaches, ranging from building the complete  
 optics on the ground to the final accuracy and mounting  
 them so as to preserve the mirror figure through the injec-  
 tion into final orbit, to providing for both alignment and  
 adjustment of individual mirror element shapes on-orbit.  
 We are considering the latter approach, since it is the most  
 general, and it mitigates a number of manufacturing,  
 spacecraft, and environmental issues.

## 5.3. Materials and fabrication

To achieve high efficiency X-ray reflection requires graz-  
 ing angles of order 1/2°. Therefore to achieve 100 m<sup>2</sup> of  
 effective area requires more than 10<sup>4</sup> m<sup>2</sup> of optic surface  
 area. To minimize the weight of such a large area con-  
 strains us to use thin-walled mirrors, ~0.1 mm thick, which  
 are nested to build up the large area. Our study concept  
 envisions thermal forming of smooth and thin glass sheets  
 onto precision figured mandrels. The glass is intrinsically  
 smooth at high spatial frequencies. The low spatial-fre-  
 quency figure errors and the coating stress induced defor-  
 mations are to be removed by on-orbit control.

## 5.4. Alignment and figure control

On the ground, the mirror elements will be nested and  
 mounted in modules. On-orbit, either automated deploy-  
 ment or active on-orbit assembly will align these modules  
 to a fraction of the resolution requirement, using a combi-  
 nation of mechanical tolerances and six-degree of freedom  
 actuators. Then the figure errors, launch deformations, and  
 thermal deformations are removed iteratively with align-  
 ment adjustment after extension of the optical bench. We  
 anticipate infrequent optic figure adjustment throughout  
 the mission, as necessitated by long term thermal drifts  
 or radiation damage to the glass. For the precise figure  
 adjustment, we are considering the use of actuators such

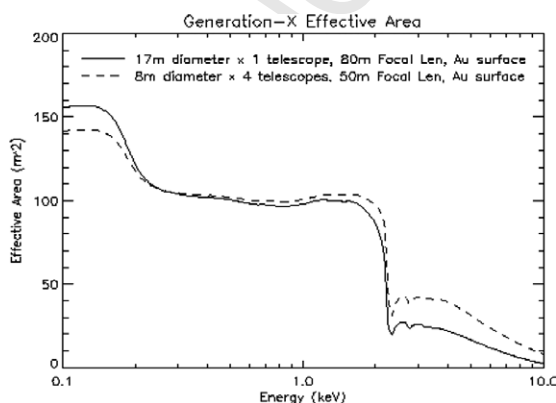


Fig. 2. Effective area for two candidate designs for Gen-X.

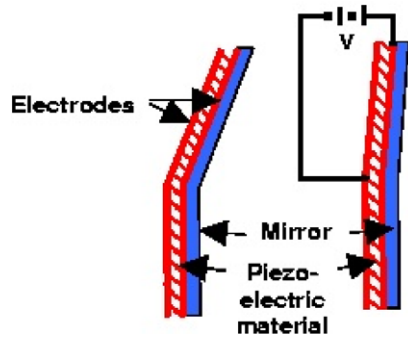


Fig. 3. Piezo-electric mirror figure control. Under an applied voltage  $V$ , the piezo material imparts a local force to the mirror (other strips would work in an orthogonal direction).

354 as micro-electro-mechanical (MEM) systems or thin film  
355 piezo-electric devices mounted on the back surface of the  
356 mirrors (Fig. 3).

## 357 6. Instruments

358 Gen-X requires X-ray detector capabilities significantly  
359 beyond those achieved to date. We need developments  
360 for high detection efficiency down to 100 eV, spectral reso-  
361 lution of  $10^3$ – $10^4$  for point and extended sources, high  
362 count-rate capability, large detector areas to provide a sub-  
363 stantial FOV, and non-X-ray background per unit detector  
364 area  $10\times$  lower than Chandra. Gen-X, like Chandra, will  
365 utilize a suite of complementary science instruments, with  
366 one chosen for any specific observation.

### 367 6.1. A cryogenic imaging spectrometer

368 A cryogenic imaging spectrometer will be a key instru-  
369 ment. It provides near perfect detection efficiency above  
370 1 keV, and provides better energy resolution than gratings  
371 over the upper range of the Gen-X energy band. Current  
372 performance is 2.4 eV resolution at 1.49 keV, so Gen-X  
373 requires only a factor of  $\sim 3$  improvement to achieve the  
374 minimum spectral resolution at 1 keV. Improved energy  
375 resolution is expected from the on-going development of  
376 transition edge sensors, super-conducting tunnel junction,  
377 or metallic magnetic or kinetic inductance devices. Position  
378 resolution of 250  $\mu\text{m}$  has been demonstrated in the labora-  
379 tory and is baselined for Con-X. This scales to 1" for Gen-  
380 X with a 50 m focal length. This device will not be required  
381 to achieve the ultimate imaging capability.

### 382 6.2. A grating spectrometer

383 A grating spectrometer will be required to provide  
384 energy resolution in the range  $10^3$ – $10^4$ , especially below  
385 1 keV. Gratings are being considered in either a transmis-  
386 sion or reflection configuration, which can draw directly  
387 upon either Chandra and XMM or Con-X experience,

respectively. Gratings will provide the highest resolution  
spectroscopy for point sources at large redshifts.

### 6.3. Imager

To exploit the 0.1" imaging, Gen-X requires a camera  
with a large field of view, broad energy response, moderate  
energy resolution, spatial resolution better than  $0''.1$ , and  
high count rate throughput. We are considering Si devices  
with active pixel readout (Holl et al., 2003; Strueder and  
Lechner, 2003), or fully depleted pn frame-store arrays.  
To date such devices have been built with 75  $\mu\text{m}$  pixels  
and  $5e^-$  readout noise. Since active pixel sensors do not  
involve charge transfer, CTI is not relevant and they are  
much less susceptible to radiation damage. They can be  
read in less than 1 ms, so that dark current does not build  
up even at relatively warm temperatures. The rapid readout  
also means that much less attenuation of external visible  
light is required, so that they can be operated without opti-  
cal blocking filters and thus achieve high quantum effi-  
ciency down to 100 eV. High time resolution allows use  
of anti-coincidence to reduce charged particle background.  
We would mosaic these detectors to provide a  $15' \times 15'$   
FOV.

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more technical discussion of Gen-X by Cameron et al.  
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